



ICAO

UNITING AVIATION



Application of Liquid Hydrogen as Aviation Fuels

Takayuki Kojima

Propulsion Systems Research Group
Institute of Aerospace Technology
JAXA

ICAO HQ, Montréal, Canada

9 – 10 SEPTEMBER 2014



ICAO | UNITING AVIATION



Characteristics of Liquid hydrogen as Aviation Fuels

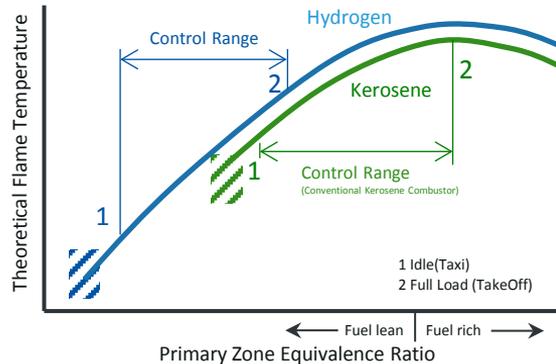


Hydrogen as Aviation Fuel (1/3)

In comparison with the current jet fuels,

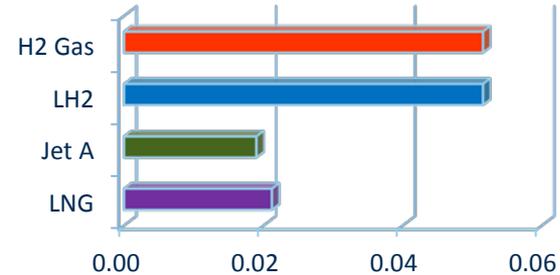
Merits

- ✓ Potentially Zero (CO₂) emission
- ✓ Higher energy content per weight(3 times) [Fig. 1]
- ✓ Potential for lower NO_x emission [Fig. 2]
- ✓ high performance as a coolant [Fig. 3]

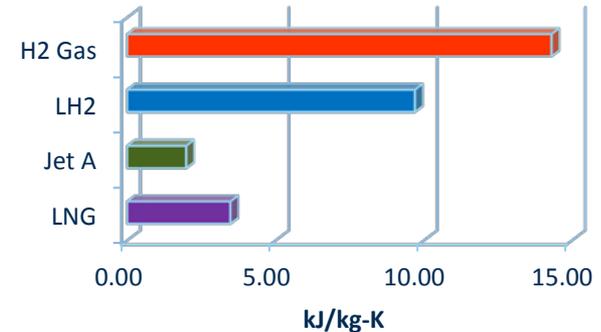


[Fig. 2] Temperature characteristics hydrogen and kerosene

[1] Ziemann J et al, EQHPP, Phase III. 0-3. Final Report,1992



[Fig. 1] Fuel properties (Heat release per weight)



[Fig. 3] Fuel properties (Specific Heat)



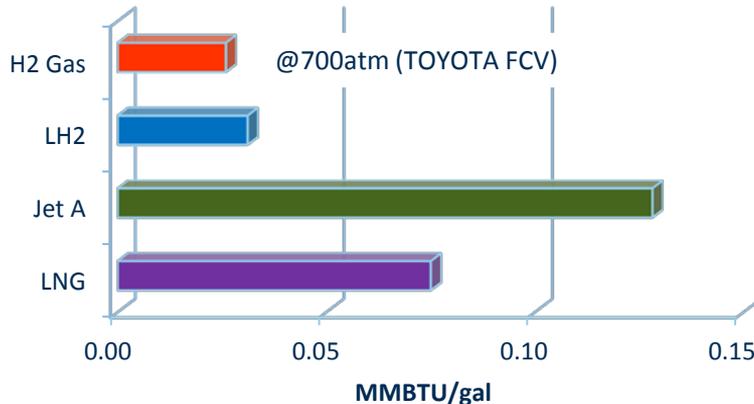
Hydrogen as Aviation Fuel (2/3)

Merits (cont'd)

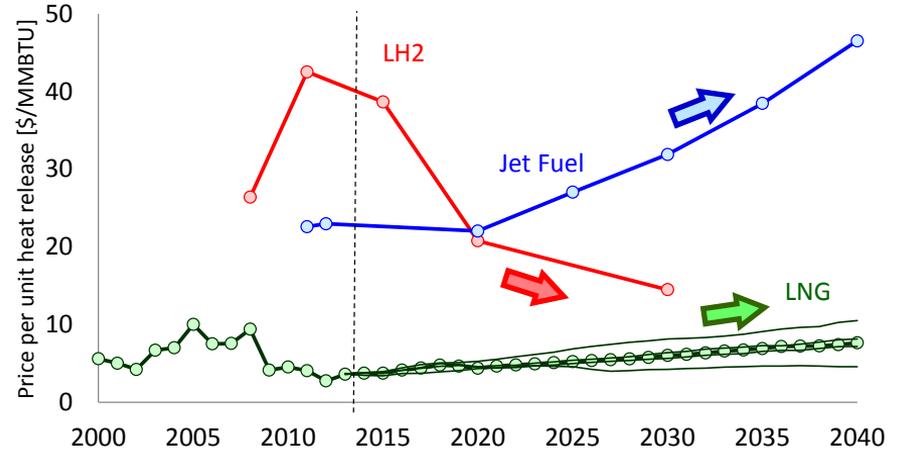
- ✓ declining production cost [Fig. 4]

Demerits

- ✓ Lower energy content per volume (1/4 times) [Fig. 5]
- ✓ Hard handling in storage and supply
- ✓ Material limitation (hydrogen embrittlement)



[Fig. 5] Fuel properties (Heat release per volume)



[Fig. 4] Production Cost Target

[2] Annual Energy Outlook 2014, U.S. EIA

[3] U.S. DOE hydrogen energy annual report 2013



Hydrogen as Aviation Fuel (3/3)

Additional Concerns

- ✓ Fuel sustenance supply (with environmental compatibility)
- ✓ Infrastructure (airport)
- ✓ Safety [Fig. 6]
 - The aviation industry is ideal to demonstrate a hydrogen fuelled transport society, since experts in restricted areas at airports and aircrafts can supply and manage the fuel.
- ✓ Impact of water vapor emission (> 2 times) on atmosphere [Fig. 7]

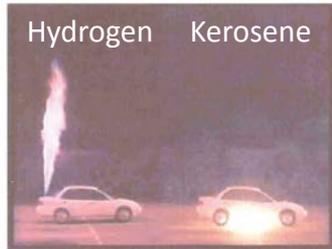
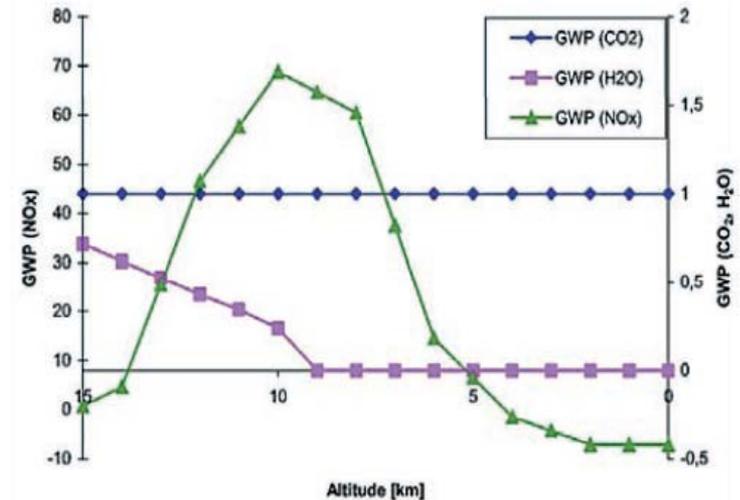


Photo 2 - Time 0 min, 3 seconds - Ignition of both fuels occur. Hydrogen flow rate 2100 SCFM. Gasoline flow rate 600 cc/min.



Photo 3 - Time: 1 min, 0 sec - Hydrogen flow is subsiding, view of gasoline vehicle begins to enlarge

[Fig. 6] Flammability comparison of hydrogen and kerosene



[Fig. 7] GWP regarding altitude

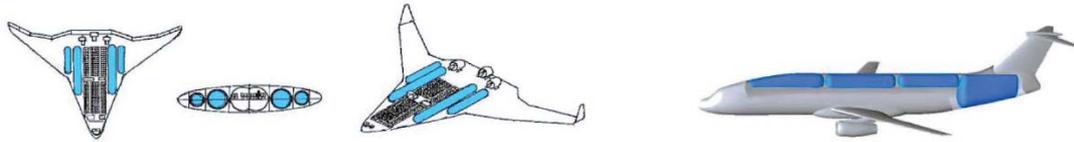
[5] Sevansson F et.al, Aerospace Science and Technology, 2004:8:307-320

[4] Westenberger A, LH2 as alternative fuel for aeronautics-study on aircraft concepts, 2006.



Aircraft and Tank configuration

- ✓ Since liquid hydrogen has low density (1/4 of Jet fuel), tank configuration becomes more important.
- ✓ By the progress of composite material technologies, hydrogen aircraft is getting closer to ultimate goal.

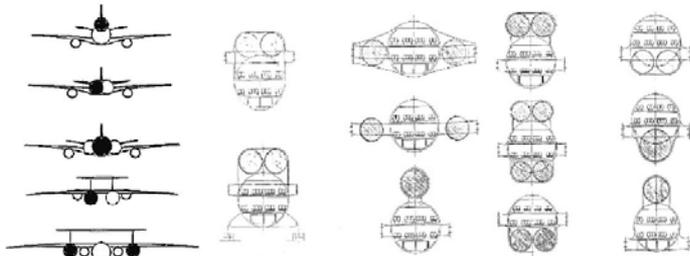


BWB concept, CRYOPLANE project

[6] Westenberger, A. et.al, Hydrogen Fueled Aircraft, AIAA2003-2880

ALT: 65,000 feet
Payload: 450 lb

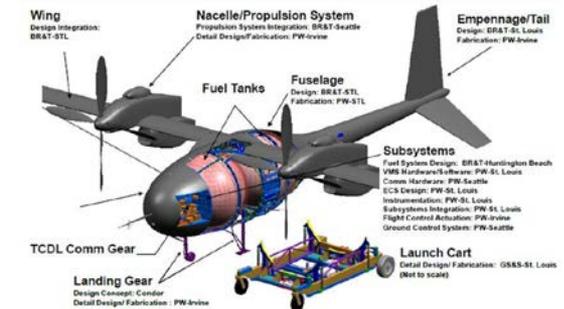
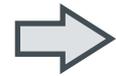
First flight: 2012



Tank Layout concepts

[7] Daggett, D. et.al, Commercial Airplanes Hydrogen Fueled Air-planes, 2003.

[Fig. 8] Conceptual Studies of Hydrogen Airplane



[Fig. 9] Boeing Phantom Eye
high altitude, long endurance liquid hydrogen-powered UAS



ICAO

UNITING AVIATION



Examples of JAXA activity



ICAO

UNITING AVIATION



JAXA's Research Initiatives for Aviation

ECAT

Environment Conscious Aircraft Technology Program



STAR

Safety Technology for Aviation and Disaster-Relief Program



Sky Frontier

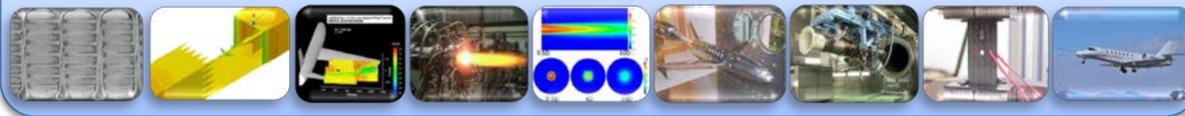
Sky Frontier Program



3 major R&D programs and Basic research

Science & Basic Tech.

Aeronautical Science & Basic Technology Research Program



Mid-term plan for 2013-2017

JAXA Proprietary



Future Image and Studies of Hypersonic Transpostation

[7] Taguchi, H. et.al, AIAA-2014-2790.



100km altitude
for Space Tourism

Sub-Orbital Plane



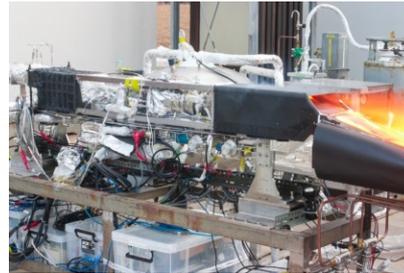
Fly Across the Pacific Ocean in 2hours



Hypersonic Transport Aircraft



Hypersonic Aircraft Technology

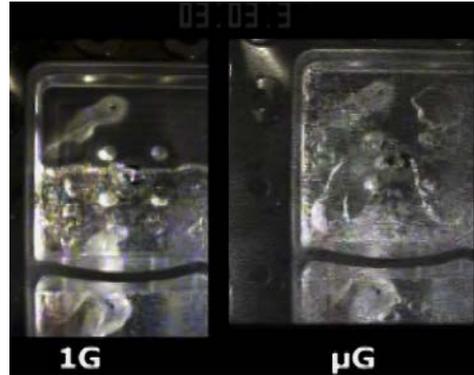


Hypersonic Propulsion Technology



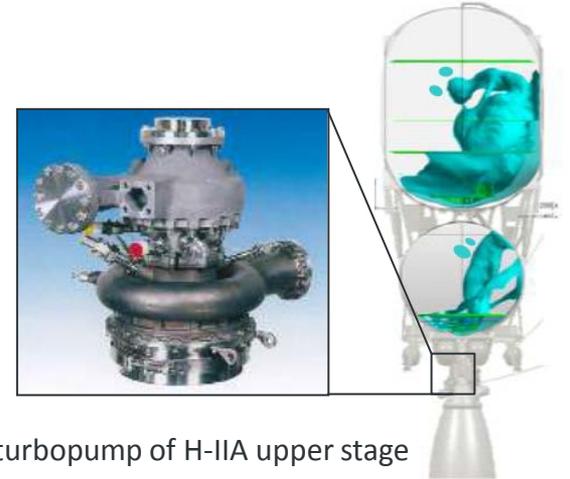
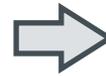


Flight Experiment of cryogenic liquid



cryogenic liquid flow under microgravity condition

JAXA's sounding rocket S-310
[04/Aug/2014, Uchinoura Space Center]



LH2 turbopump of H-IIA upper stage

- JAXA has conducted flight experiment of sounding rocket S-310 to measure behavior and heat transfer properties of cryogenic two phase flow under microgravity condition.
- Results will be applied to improve chill-down operation of second stage for JAXA's main rocket H-IIA.

[7] JAXA Press release, 5/Aug/2014

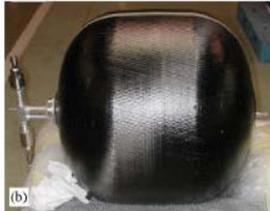
[8] JAXA brochure, "H-IIA upgrade".



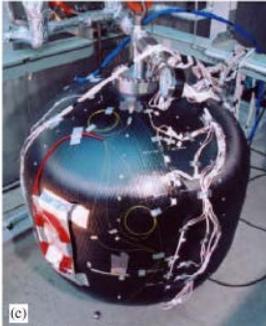
Development of light weight LH2 feed system



(a) Aluminum Liner and Mandrel



(b) FW CFRP over the Liner



(c) Strain and Temperature Sensors



(d) PUF Insulation and Water-resistant Taping



Test setup

Superconducting Turbopump experiment



Rotor of Superconducting Motor with MgB2

[10] Kyushu university Press release, 10/May/2012 (in-Japanese).

Development of metal-lined CFRP cryogenic tank

[9] Higuchi, K et.al, Acta Astronautica 57 (2005) 432 – 437.

- Light weight CFRP cryogenic tank is a key technology. JAXA have been researching material-lined CFRP tank.
- Superconducting turbopump as one aspect of MEA (more electric aircraft) would be one solution to simplify pressurization system of LH2 aircraft.



Summary

- Hydrogen has been a long-term future solution to environment concerns and energy dependency.
- Research on hydrogen fuelled aircraft has been conducted quite a long time, we are getting closer to ultimate goal.
- Recent activities for hydrogen society is a good background to accelerate the development of hydrogen fueled ground and aircraft sytem.