

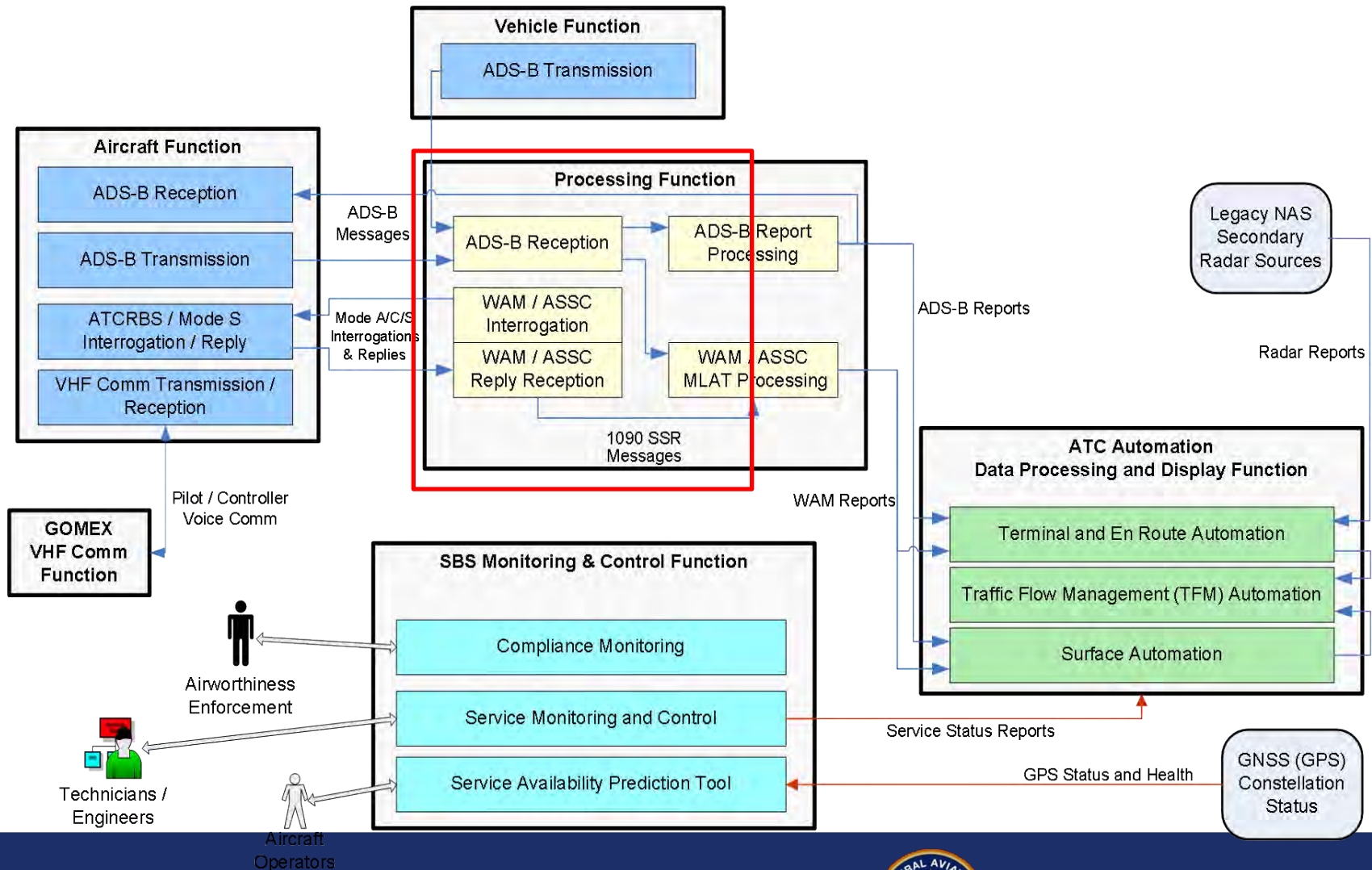
# Detailed Overview of Ground Systems



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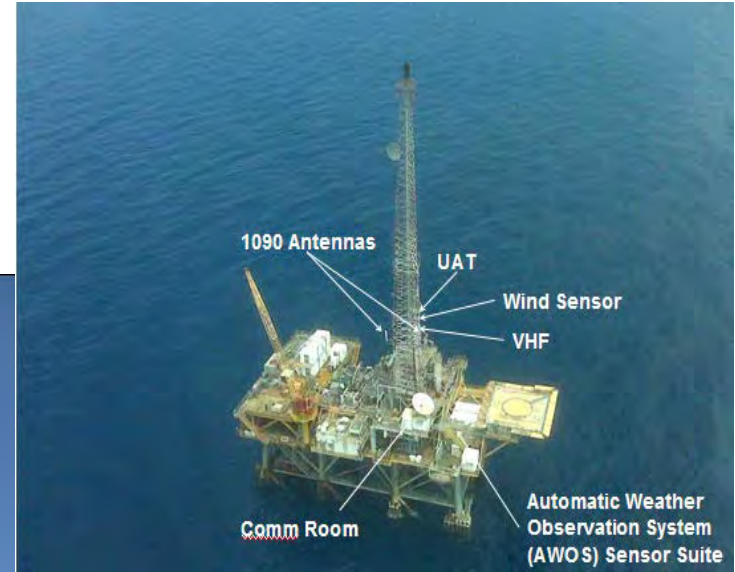
# FAA Surveillance Functional Architecture



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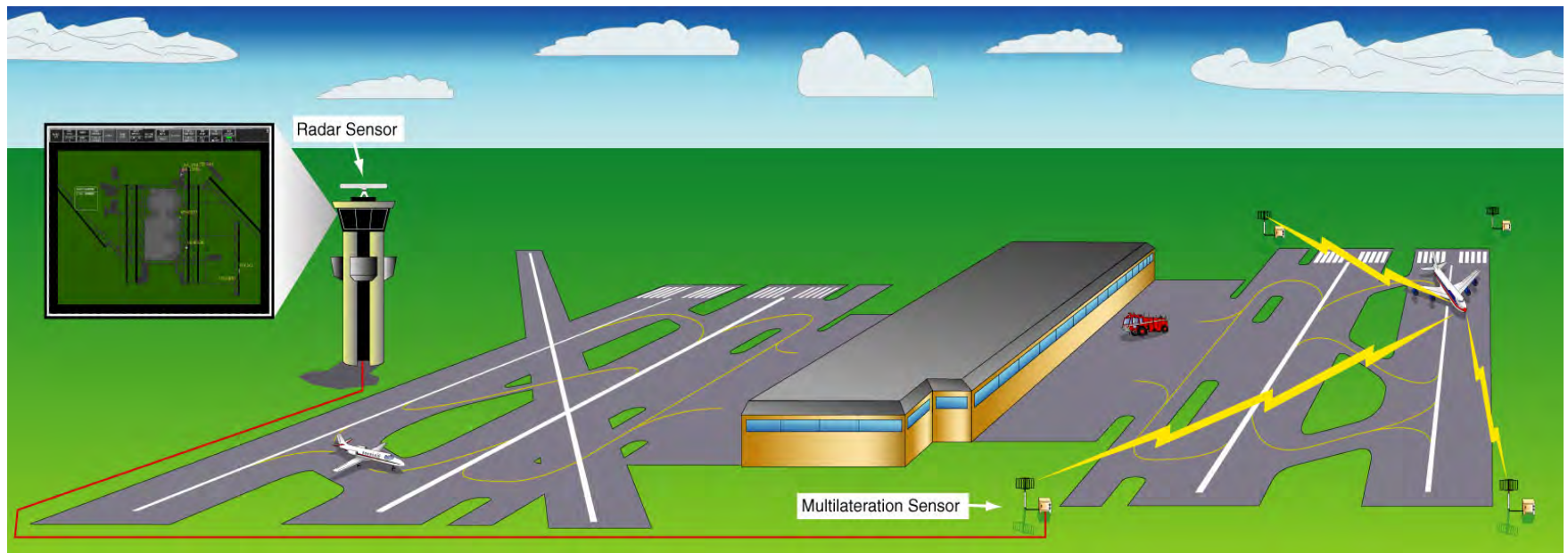


# FAA ADS-B Ground Systems



# ASDE-X Purpose

- **Airport Surface Detection Equipment - Model X (ASDE-X)** provides multi-sensor surface surveillance with aircraft and vehicle identification



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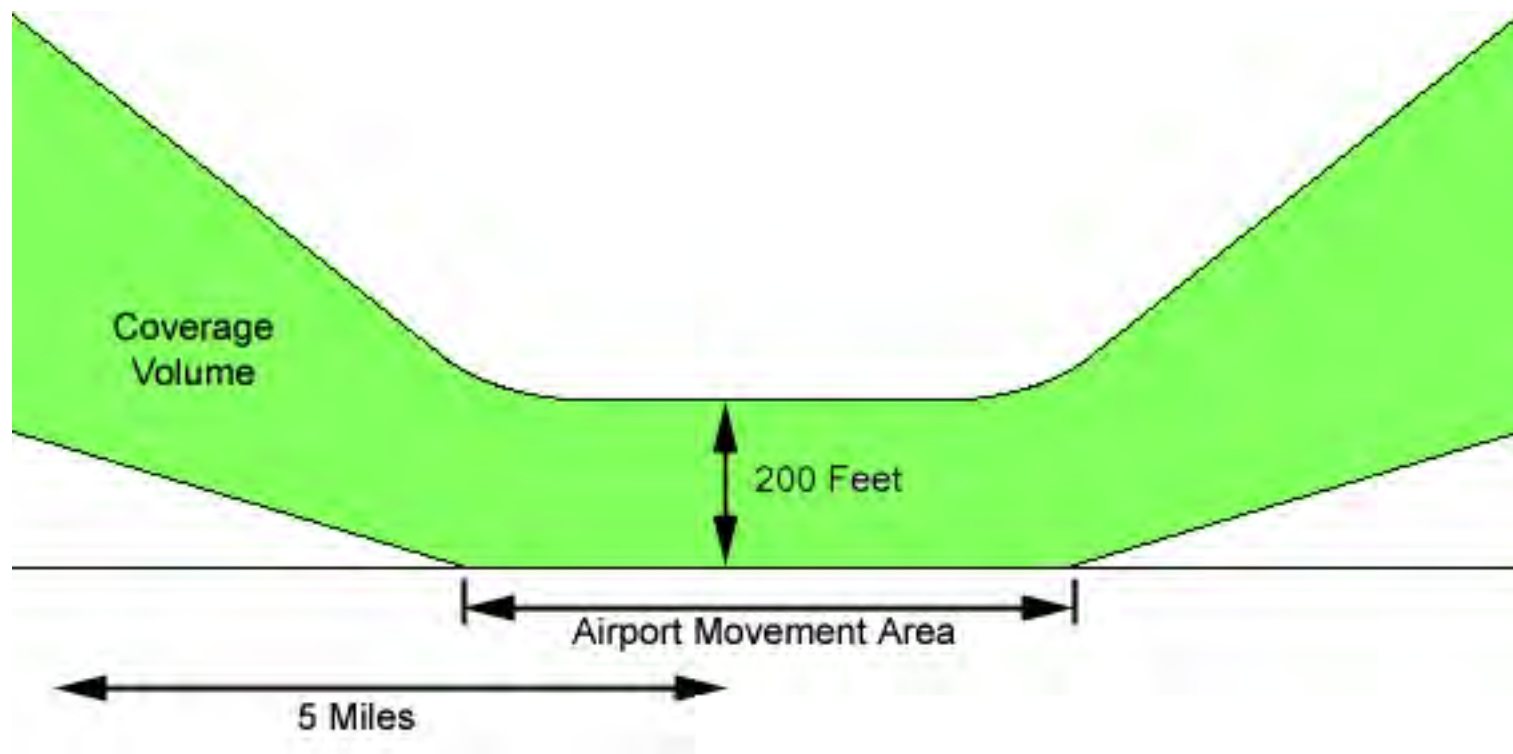
# ASDE-X Purpose

- **System Capabilities**

- Provide coverage for airport surface and approach areas, out to five nautical miles
- Track over 200 combined real surface and approach targets, with a peak initiation rate of 20 new targets per second
- Provide accurate identification of all aircraft on the ground
- Once a second update rate
- Quick track initiation, within five seconds after entering coverage area
- Integrate flight plan information
- ADS-B target reporting



# ASDE-X Coverage Volume



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# ASDE-X Subsystems

- **Multilateration (MLAT)**
  - “Secondary Surveillance System” that provides position and identification of all transponder equipped aircraft and vehicles
- **Surface Movement Radar (SMR)**
  - “Primary Surveillance System” used to detect surface targets
- **Multi-Sensor Data Processor (MSDP)**
  - Combines all sensor reports for a target, determines target position and tracks the target
- **Tower Display**
  - Provides the data and displays for Air Traffic Controller and Airway Facility Technician use



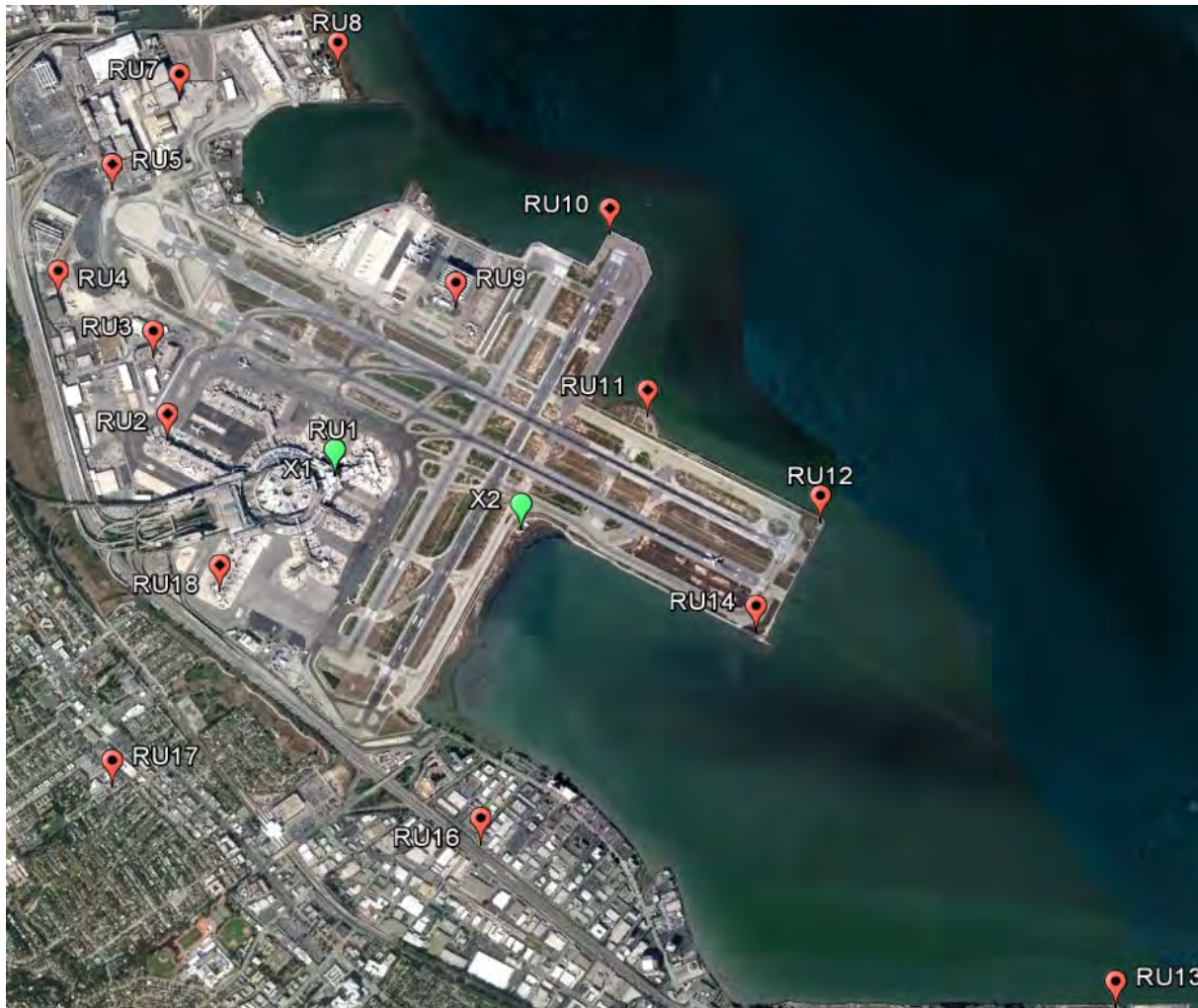
# ASDE-X Subsystems

- **Remote Monitoring System (RMS)**
  - Provides the data and displays the Airway Facility Technician uses to monitor, troubleshoot and manage ASDE-X
- **Communications**
  - Provides communications and protocol conversion between subsystems and other information sources such as ASR-9 and ARTS





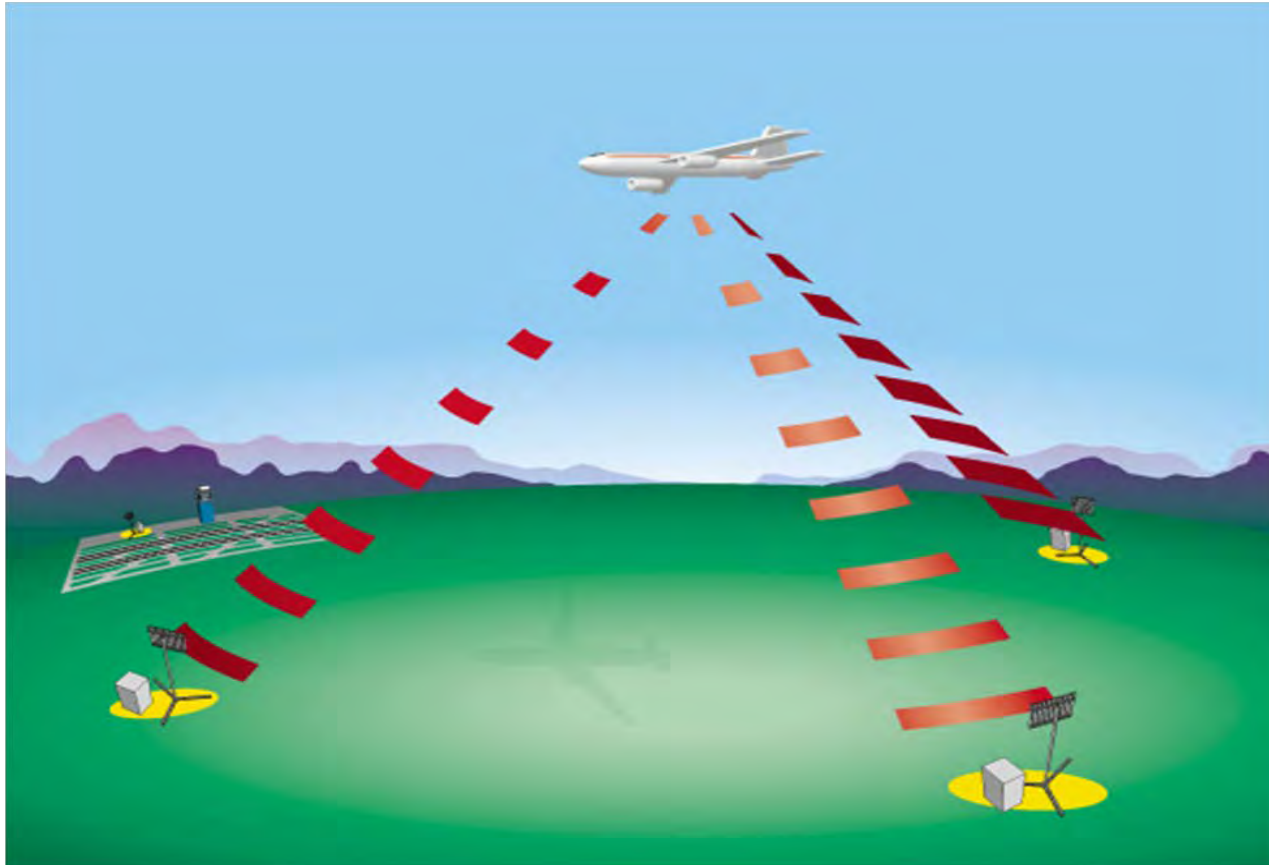
# MLAT Surface Implementation (SFO)



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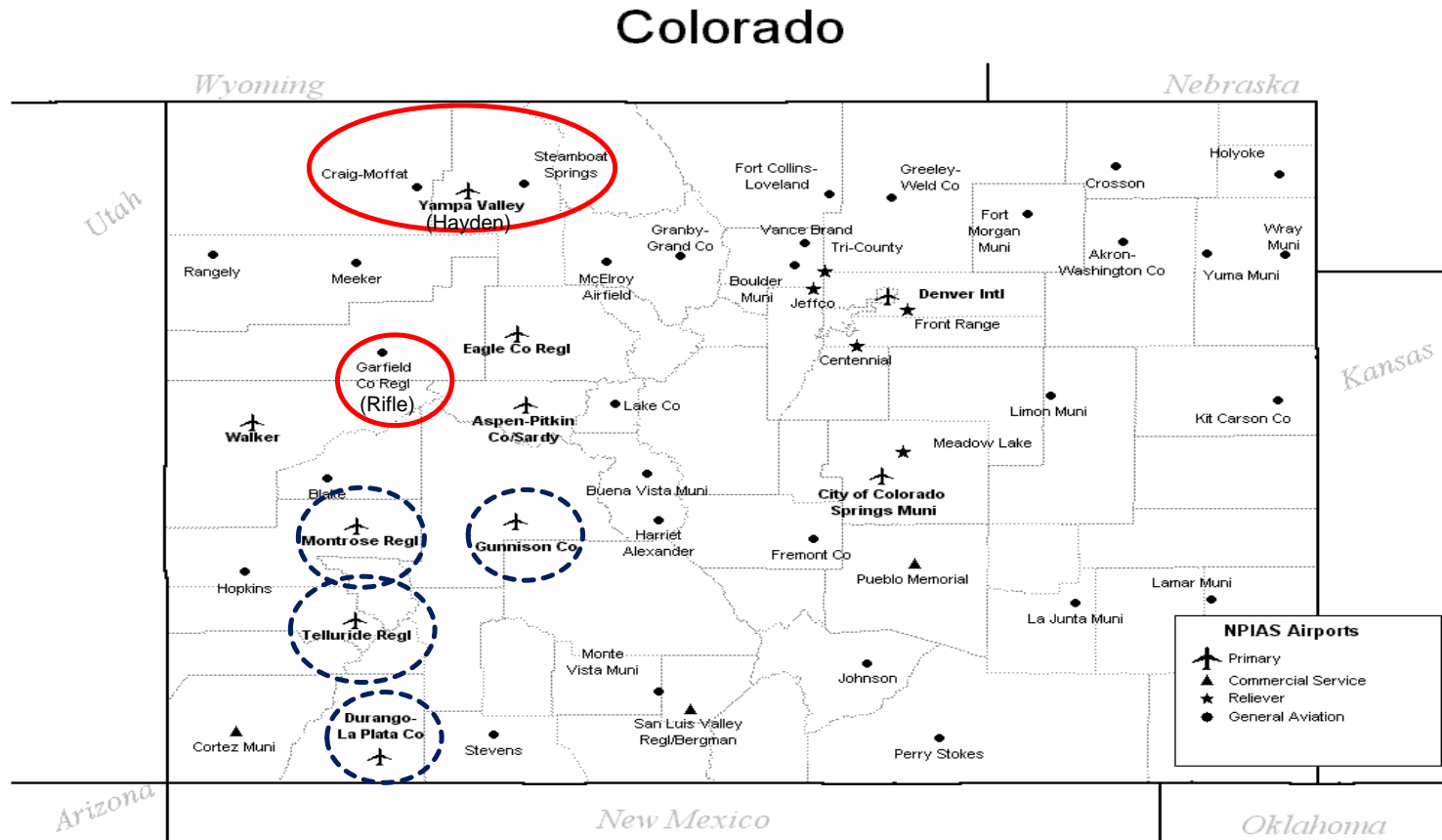
# Purpose of WAM

- WAM provides secondary surveillance with target identification



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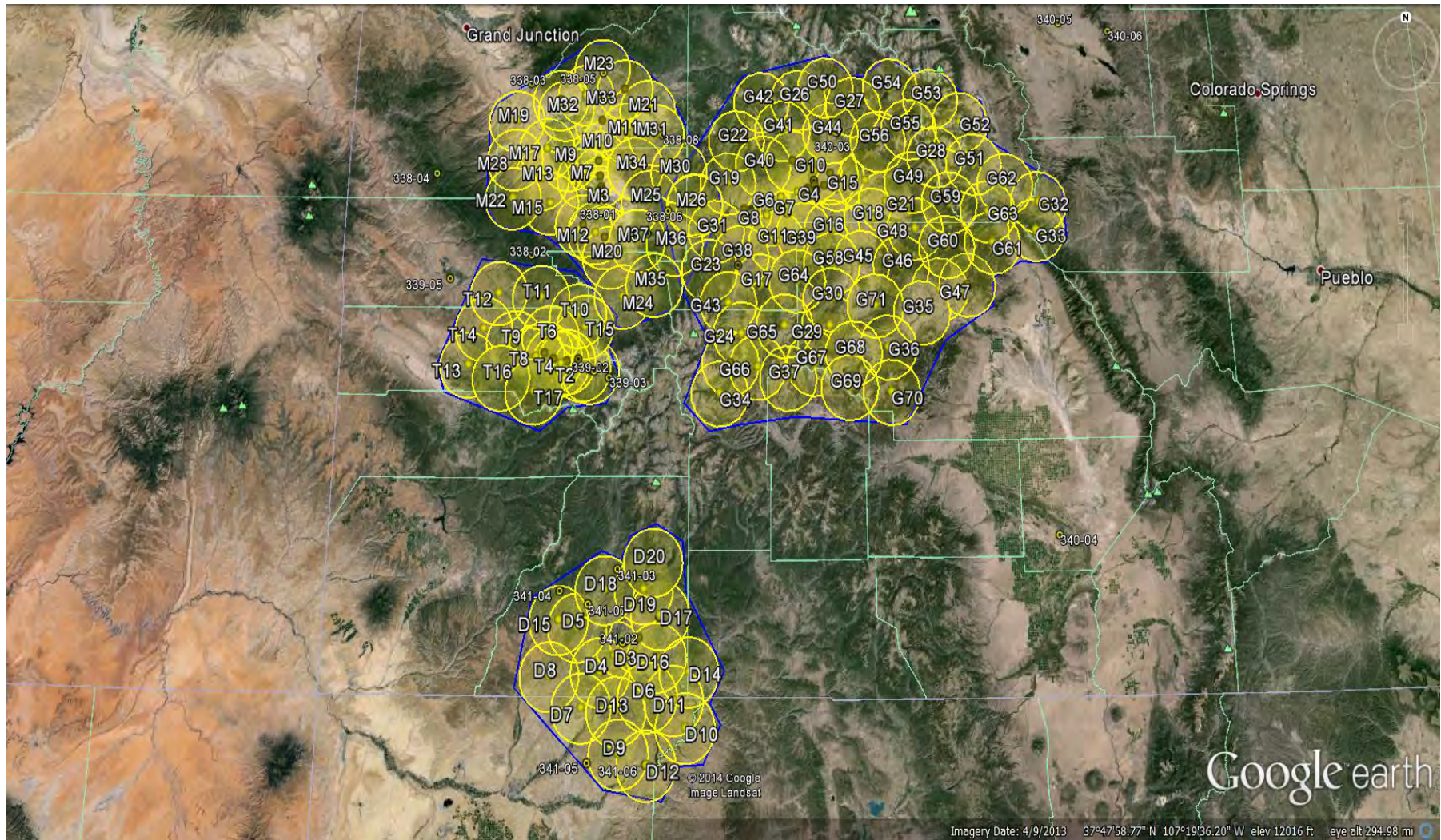
# WAM Implementation (Colorado)



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# WAM Implementation (Colorado)



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# Multilateration Theory

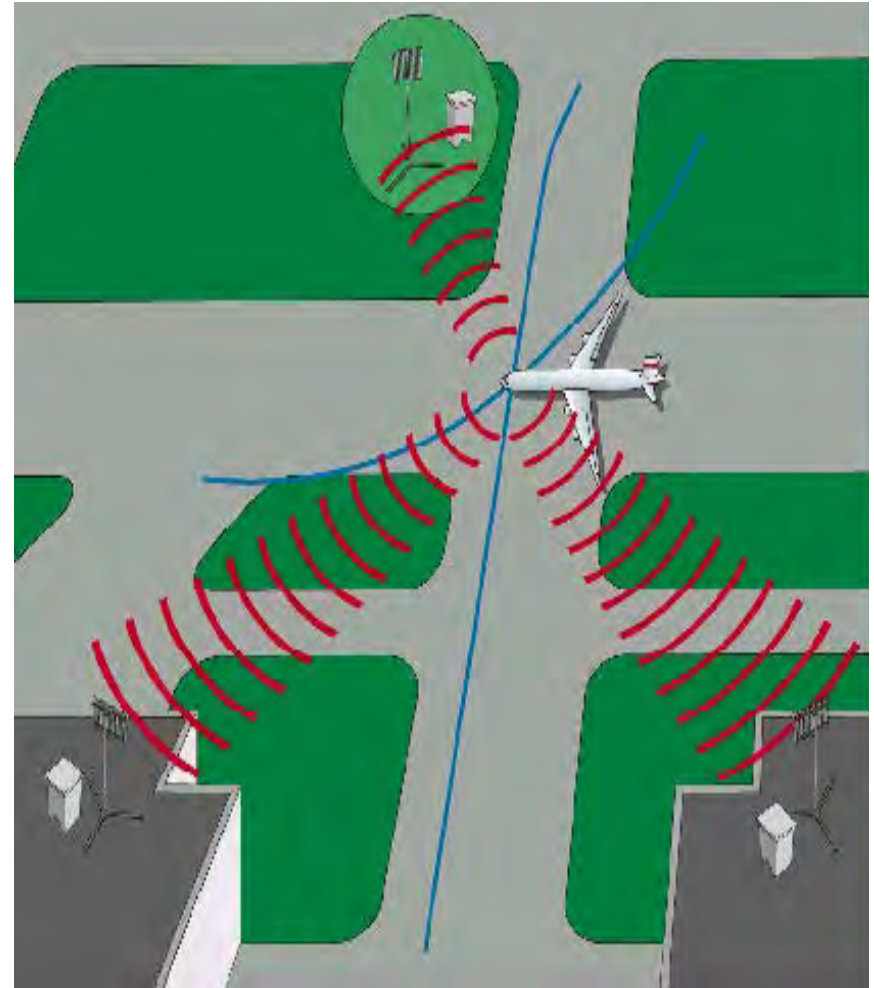
## Multilateration Concept



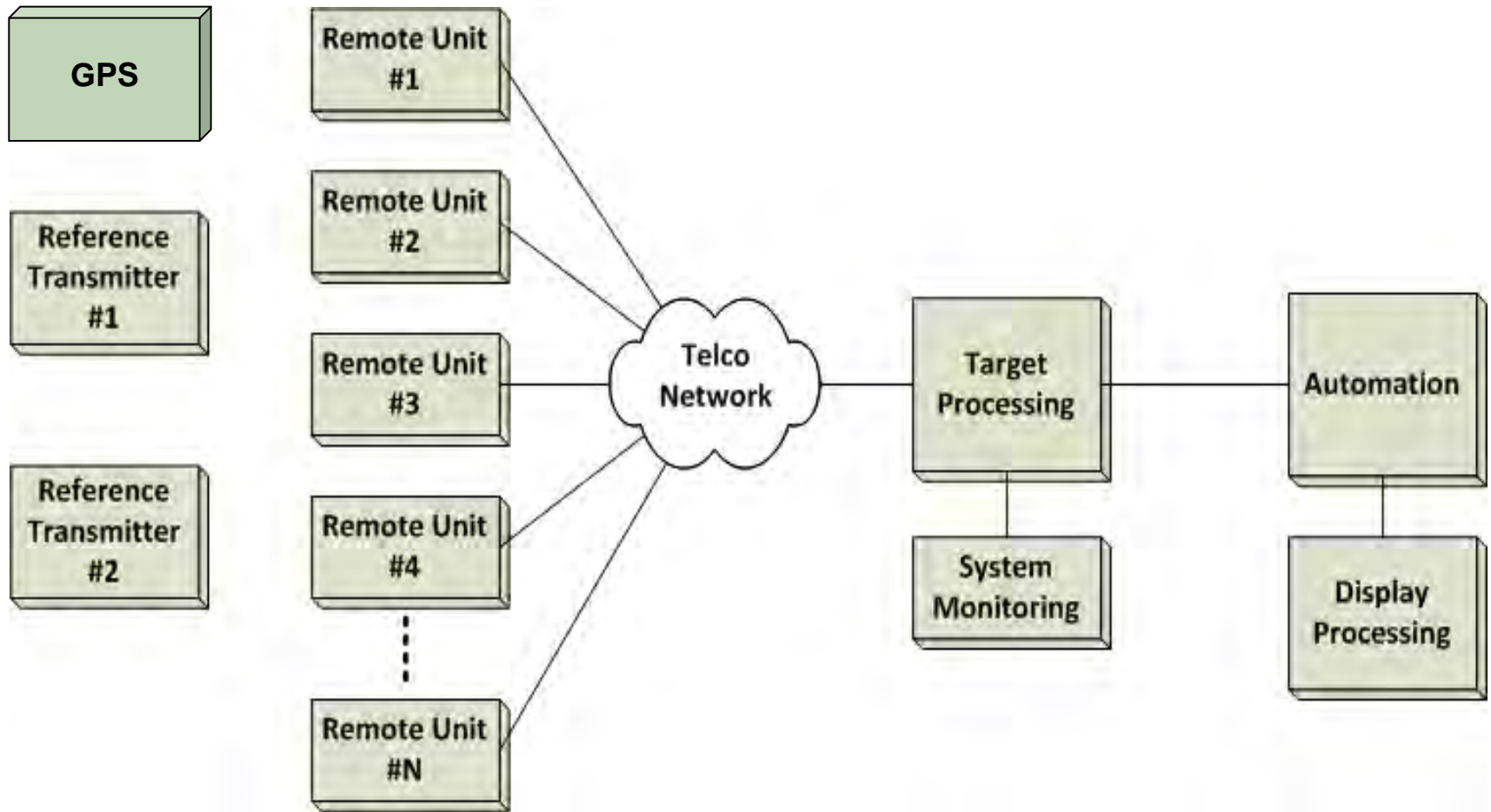
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# Multilateration Concept

- **Multilateration is based on Time Difference of Arrival (TDOA) processing**
- **The key concepts of a TDOA system include the following:**
  - RF signal is received at multiple Remote Units (RUs). Note: Minimum 3 RUs are needed to track target.
  - Signal is time-stamped at each RU
  - Time stamped signal is sent from each RU to a Target Processor (TP)
  - TP calculates the difference in signal arrival time from one RU to all other RUs and generates solution arcs
  - TP provides an estimated target (transponder) location at a theoretical intersection of solution arcs



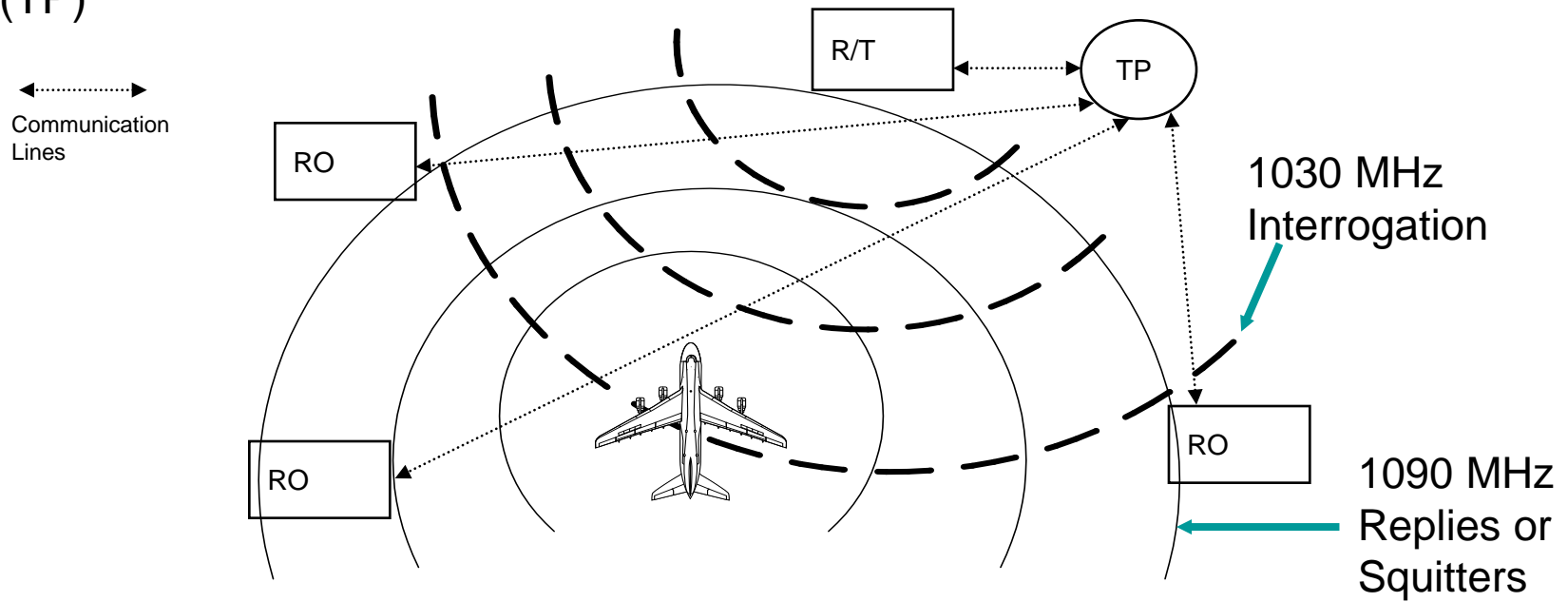
# Multilateration Architecture



# Multilateration Concept

- **WAM Multilateration Implementation:**

- Uses aircraft transponder beacon signals
- Receiver / Transmitter (R/Ts) interrogate aircraft
- Receive Only (ROs) and Receiver / Transmitter (R/Ts) receive replies and squitters
- Replies and squitters are time stamped and sent to a Target Processor (TP)





# Multilateration Theory

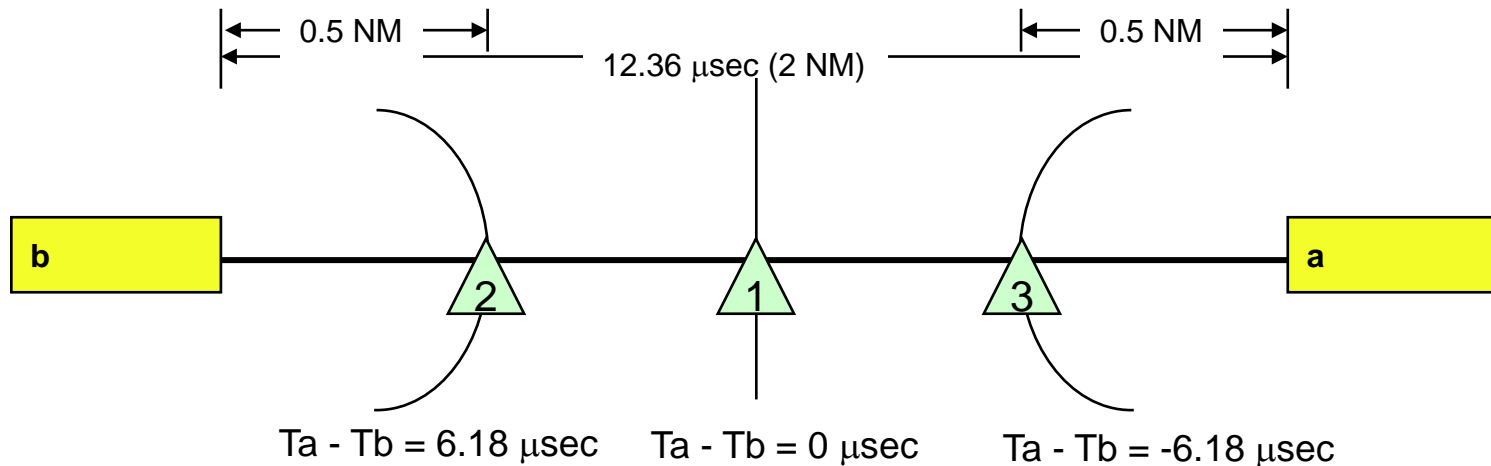
## Time Difference of Arrival (TDOA)



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# Time Difference of Arrival (TDOA)

- **Calculating the difference in a signal arrival time between two sensors.**
  - Time = Distance
    - RF energy travels at the speed of light
    - RF energy travels 1 NM in 6.18  $\mu\text{sec}$
  - $\text{TDOA} = T_a - T_b$ 
    - The difference in arrival time can be plotted on a hyperbolic arc
    - Every point on the arc must maintain this difference

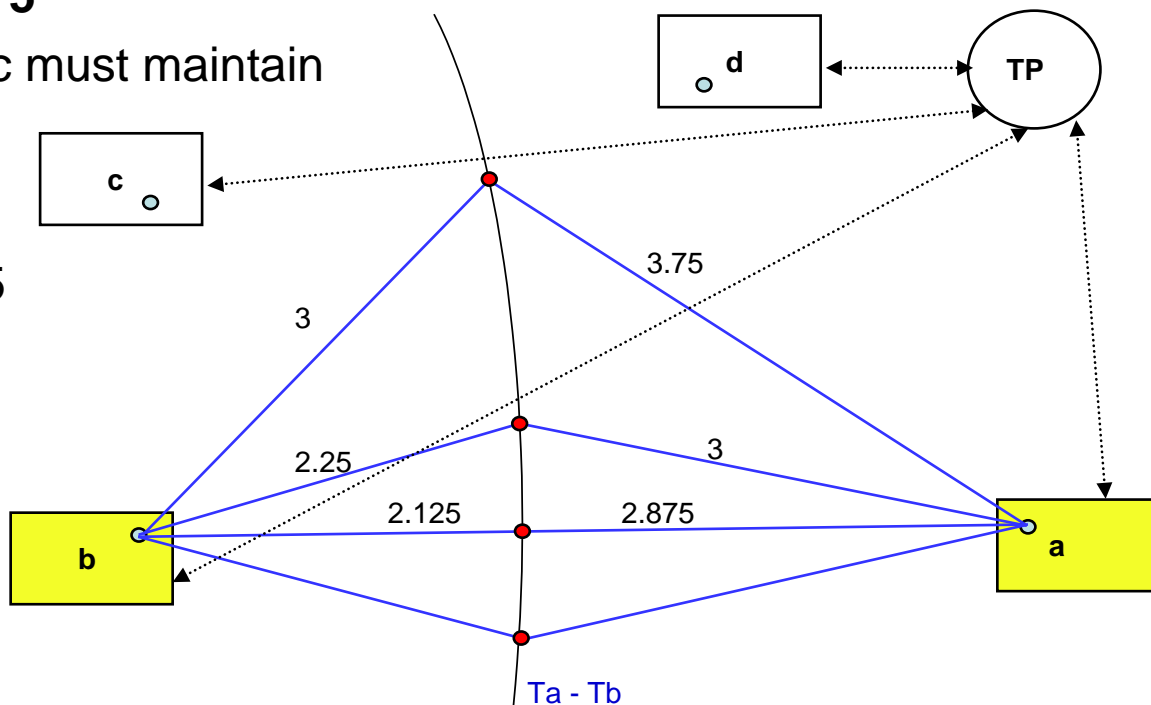


# Time Difference of Arrival (TDOA)

- **Target Processor (TP) performs TDOA calculations to generate hyperbolic solution arcs**

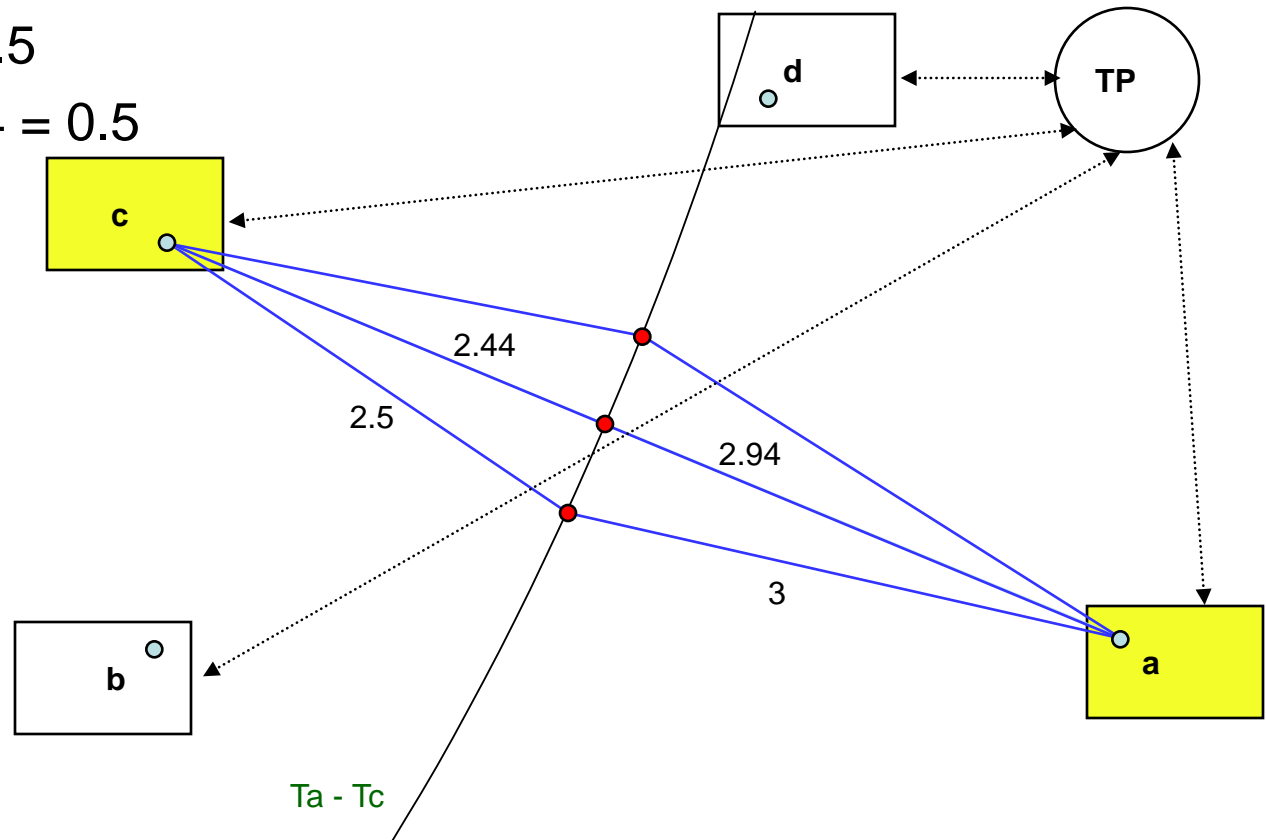
- Transponder must exist somewhere on the arc
- $\text{TDOA} = (\text{Time received at RU a}) - (\text{Time received at RU b})$
- $\text{TDOA} = T_a - T_b = 0.75$
- Every point on the arc must maintain this time difference

- $2.875 - \underline{\hspace{1cm}} = 0.75$
- $3 - \underline{\hspace{1cm}} = 0.75$
- $\underline{\hspace{1cm}} - 3 = 0.75$



# Time Difference of Arrival (TDOA)

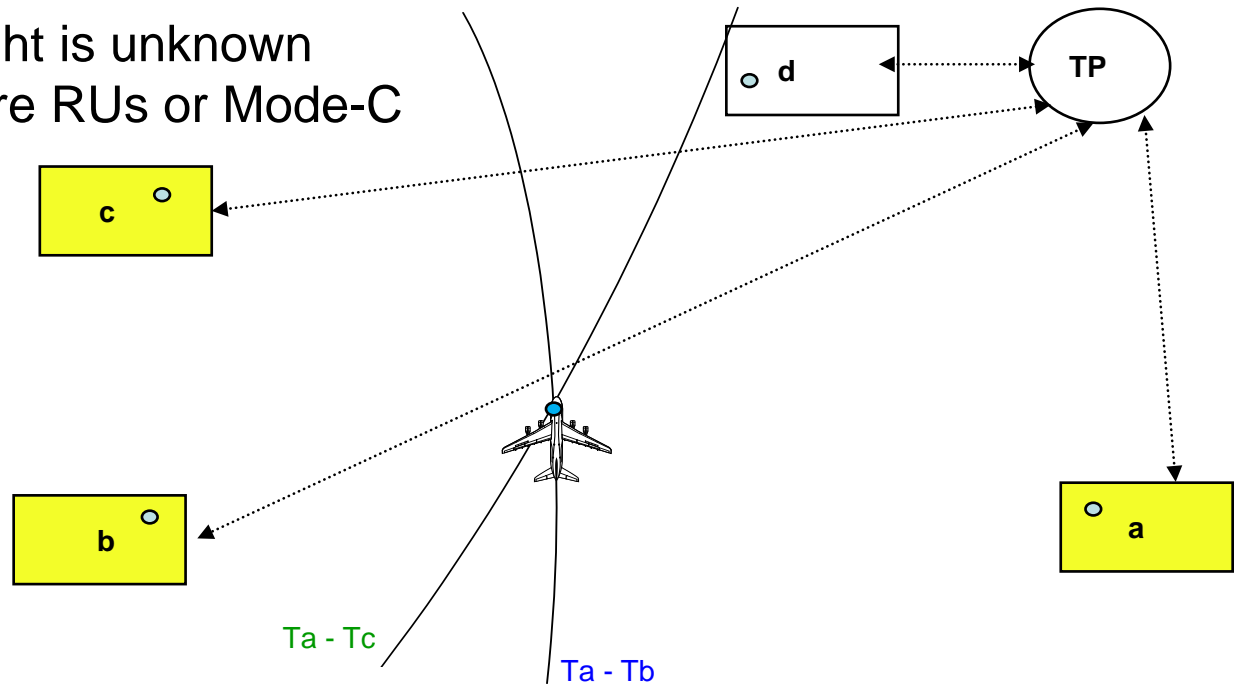
- **By itself, one arc provides no useful information.**
  - Calculating second solution arc for RU a to RU c
  - $\text{TDOA} = T_a - T_c = 0.5$ 
    - $3 - 2.5 = 0.5$
    - $2.94 - 2.44 = 0.5$





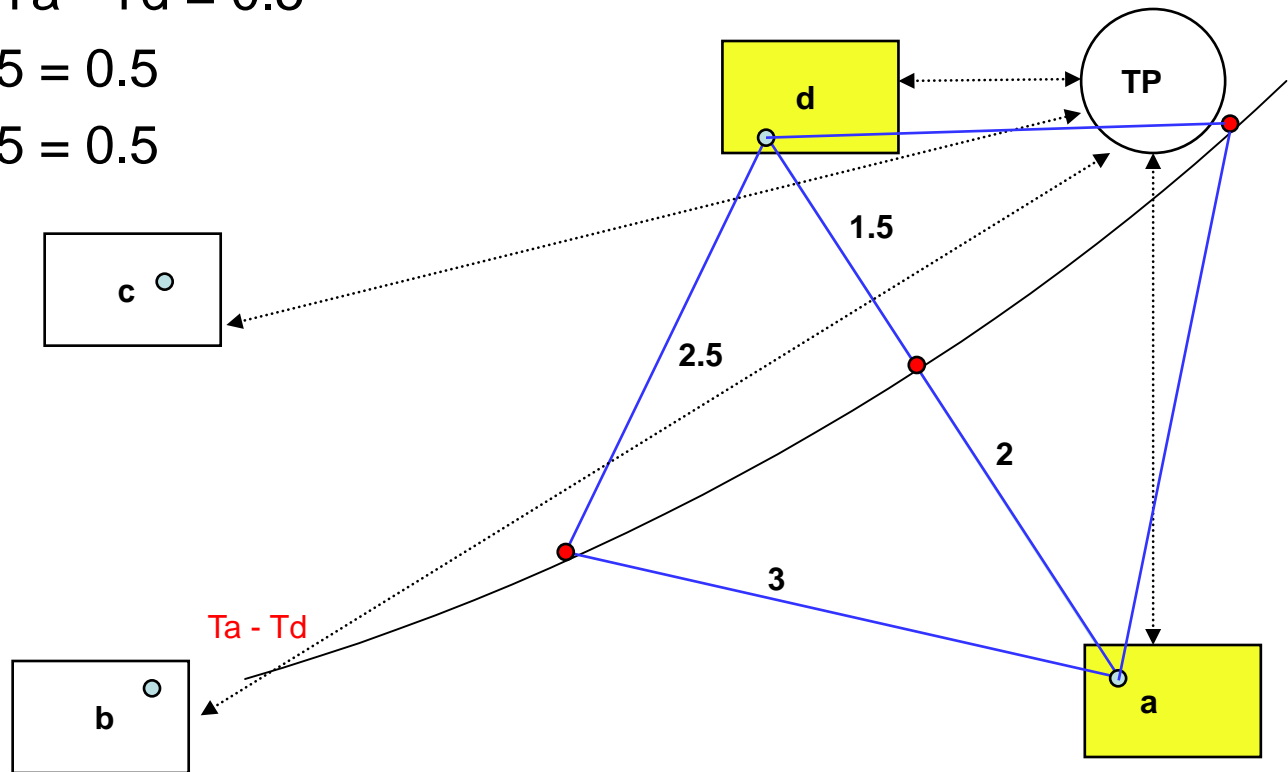
# Time Difference of Arrival (TDOA)

- To get a two dimensional point, three RUs are required.
  - 3 RUs develop intersecting hyperbola, 2D formula is used (no Z axis)
  - 2d position is generated
    - Gives range and azimuth
    - Target height is unknown without more RUs or Mode-C



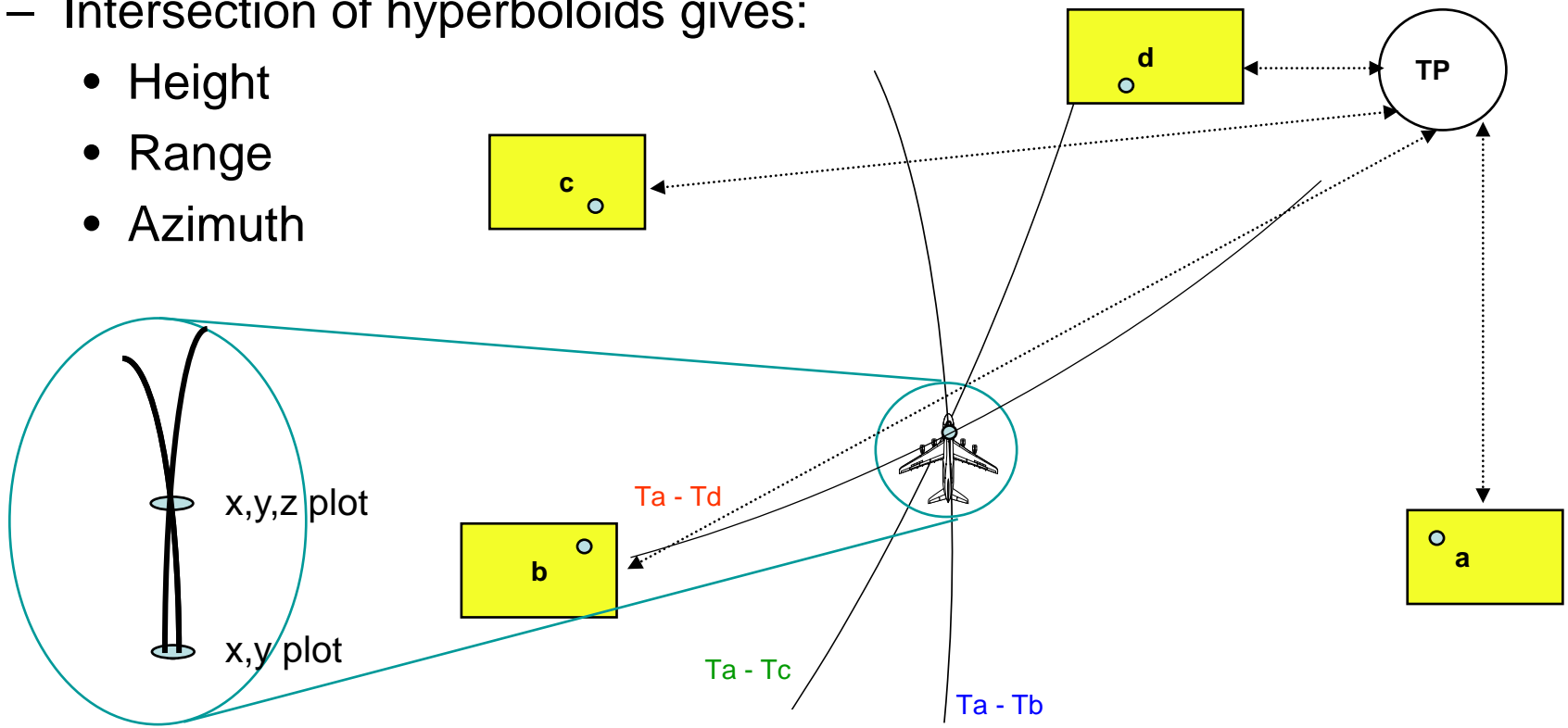
# Time Difference of Arrival (TDOA)

- To get a three dimensional point, four or more RUs are required.
  - Calculating solution arc for RU a to RU d
  - $\text{TDOA} = T_a - T_d = 0.5$ 
    - $3 - 2.5 = 0.5$
    - $2 - 1.5 = 0.5$



# Time Difference of Arrival (TDOA)

- With four or more RUs, 3D formula is used
  - Z axis is calculated
  - Generates hyperboloids (3D hyperbola)
  - Intersection of hyperboloids gives:
    - Height
    - Range
    - Azimuth



# Time Difference of Arrival (TDOA)

- **WAM systems are capable of using GPS as the primary timing source**
- **If GPS timing is not primary or unavailable, TDOA processing can be performed in the TP using Fine Time Stamps from the Remote Units (RUs)**
  - Each RU has a free running clock
    - Timestamp resolution will affect the accuracy of the solution
    - The free running clocks in the RUs are not synchronized
  - Multilateration is not possible unless the RU clocks are corrected to a common time basis
- **Reference Transmitter (RefTran) provides a known signal to calibrate the TP TDOA function**
  - RefTrans squitter once a second
  - TP knows:
    - RefTran position and Mode S identification
    - RU positions
    - Expected TDOA between all RUs for each RefTran squitter



# Time Difference of Arrival (TDOA)

## – TDOA Calibration

- Clocks in the RUs have no common time reference
- TP calculates TDOA on the RefTran squitter
- TP initiates TDOA calibration upon receipt of the RefTran's unique Mode-S identification
- TP uses the difference between the expected RefTran TDOA and the calculated TDOA to calculate RU Offsets
  - RU offsets are used to provide a common time base required for Multilateration to take place
- TP also calculates Drift rates
  - Drift rates account for differences in the RU clock circuits (clock width, speed, skew, etc...)
  - TP uses Drift rates to accurately predict changes in the RU Offset while the RefTran is not squitting
  - Drift rates improve multilateration accuracy during the 1 second period between RefTran squitters

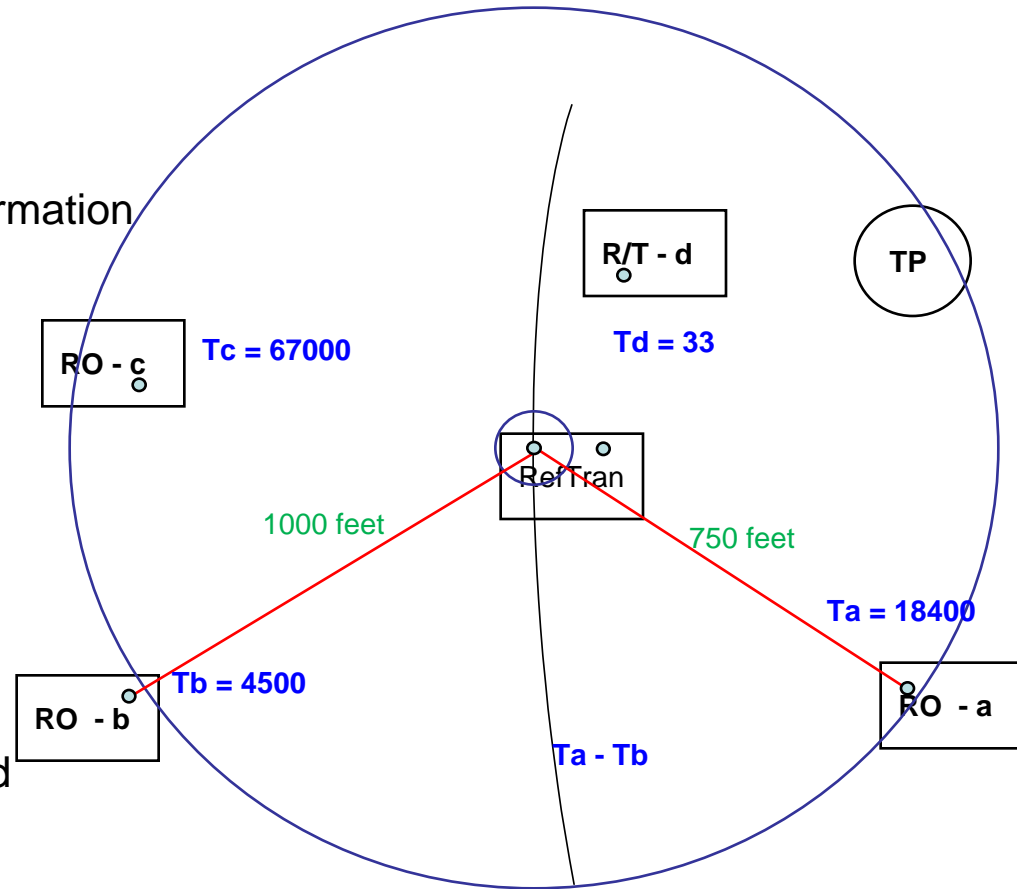


# Time Difference of Arrival (TDOA)

- **RefTran Example:**

- TP uses one RU as a reference and calculates the TDOA for all RUs
- RU counters provide 31 bits of information
- 1 clock tick = 10 nsec
- Example:

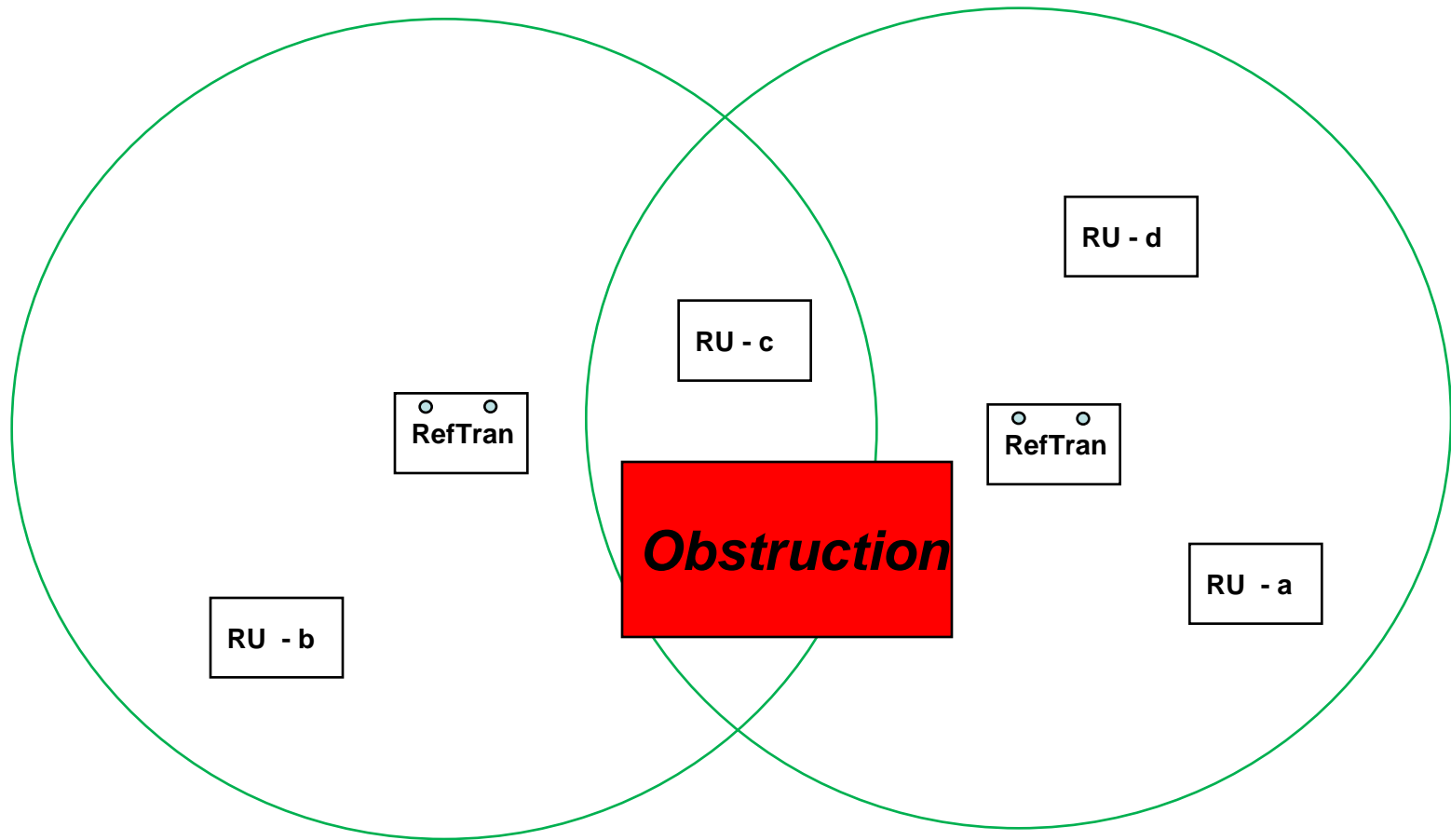
- Expected TDOA  
 $T_a - T_b = -250$  feet  
 $-250 \text{ feet} = -250 \text{ nsec}$   
 $-250 \text{ nsec} = -25 \text{ tics}$
- Calculated TDOA  
 $T_a - T_b = 18400 - 4500$   
 $T_a - T_b = 13900 \text{ tics}$
- Offset = Calculated – Expected  
 $\text{Offset} = 13900 - (-25)$   
 $\text{Offset} = 13925 \text{ tics}$
- Next return, Drift rates can be calculated



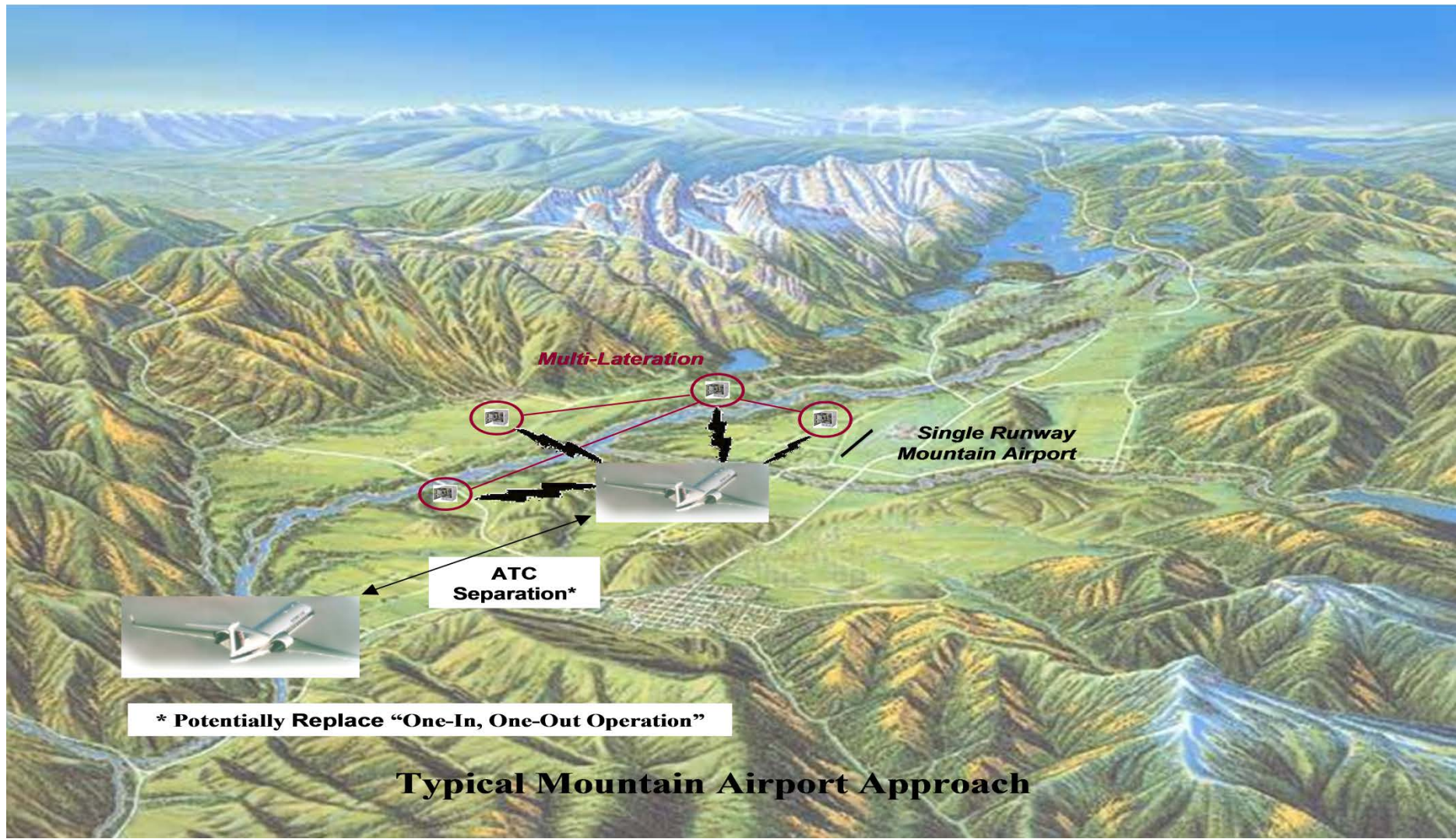


# Time Difference of Arrival (TDOA)

- RefTran must be seen by more than one RU for TDOA calibration



# Example of MLAT Implementation



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# Multilateration Theory

## Target Detection and Tracking



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# Target Detection and Tracking

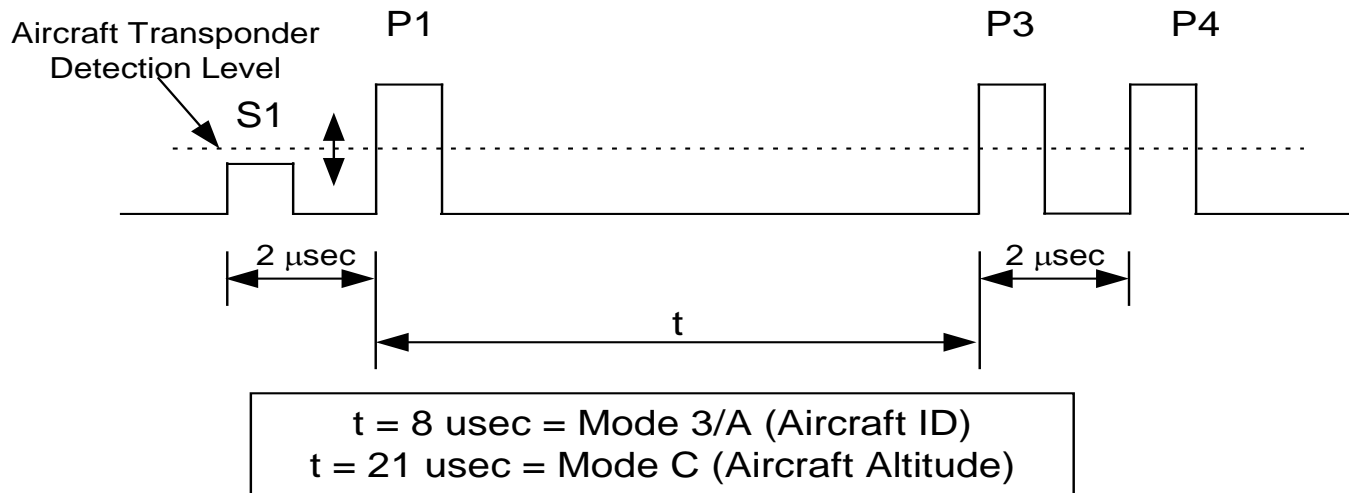
- **The WAM system performs target detection and tracking on ATCRBS, Mode S, and ADS-B signals.**
- **It can receive, decode and determine an accurate TDOA for:**
  - Air Traffic Control Radar Beacon System (ATCRBS) replies
  - Mode S (Select) signals (squitters and replies)
  - Automatic Dependent Surveillance - Broadcast (ADS-B) signals



# ATCRBS

- **WAM ATCRBS Interrogation**

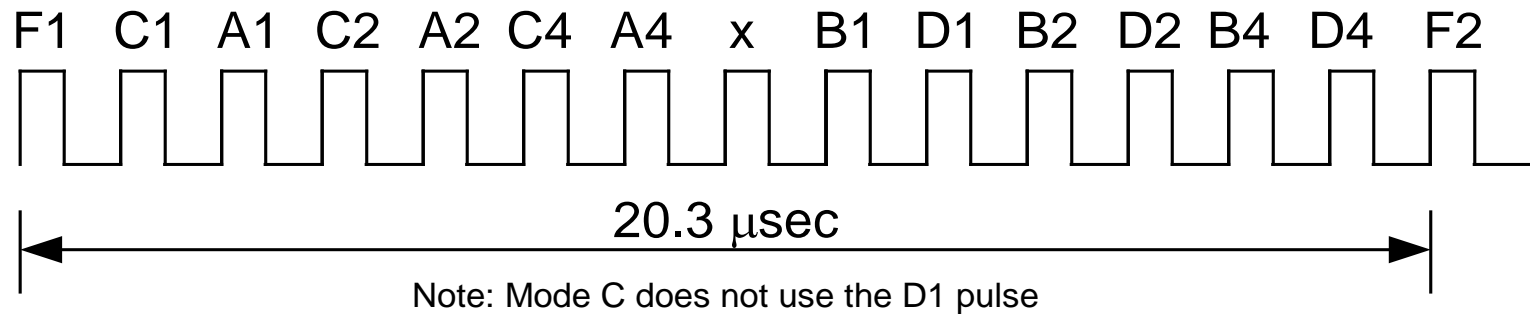
- P1 and P3 indicate Mode (3/A or C), they are  $0.8\ \mu\text{sec}$  in width and either 8 or  $21\ \mu\text{sec}$  apart.
- P4 trails P3 by  $2\ \mu\text{sec}$  and is also  $0.8\ \mu\text{sec}$  in width. ATCRBS transponders will not detect P4. Mode S transponders will decode a  $0.8\ \mu\text{sec}$  P4 as an ATCRBS only All-Call request (suppressing the Mode S transponder).
- S1 precedes P1 by  $2\ \mu\text{sec}$  and is used to simulate a P1-P2 suppression pulse.



# ATCRBS

- **ATCRBS Replies**

- An ATCRBS reply provides aircraft identification(3/A) or altitude(C) information
- ATCRBS transponders reply with either their Mode 3/A or C code using a 12 bit reply bracketed by 2 framing pulses
- Framing pulses are 20.3  $\mu$ sec apart
- 12 bit reply consists of four 3 bit numbers, A B C D, decoding provides the reply code





# ATCRBS Target Acquisition

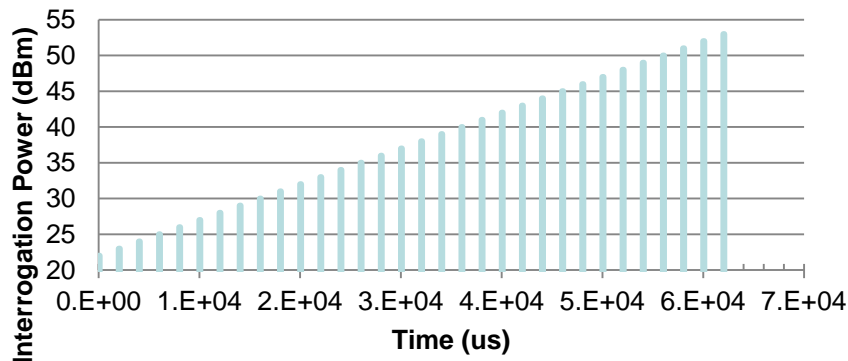
- **ATCRBS targets are acquired by using Mode A/A doublet interrogations in a Whisper/Shout Sequence**
- **Doublet Interrogations**
  - A doublet is two interrogations with a close fixed spacing (100s of microseconds)
  - A Mode A/A doublet is two Mode A all call interrogations with close fixed spacing
  - The doublet spacing is fixed at each interrogator within a service volume and allows the replies to be filtered at the receivers to reduce the impact of FRUIT and only deliver relevant replies



# ATCRBS Target Acquisition

- **Whisper / Shout (W/S) Sequence (Acquisition)**
  - Series of range-limited interrogations with incrementing power levels
  - Analogous to glancing at darts on a dartboard and identifying groups of darts by their ring number
  - Sequence has low repetition rate ( $\sim 5$  s) at each radio ( $\Rightarrow$  over multiple radios the effective repetition rate is  $\sim 2.5$  s)

Example mode 3/A interrogation sequence with 32 steps & 2 ms spacing



# ATCRBS Target Tracking

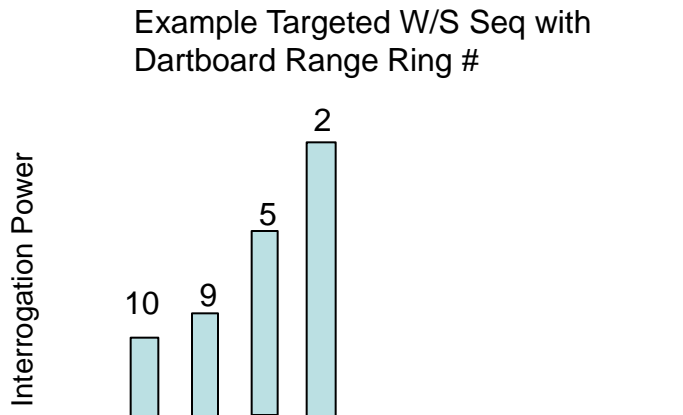
- **ATCRBS Targets are tracked by using AC doublet interrogations in a Whisper/Shout Sequence**
- **Doublet Interrogations**
  - Doublet Interrogations are similar to the ATCRBS acquisition doublets, but are Mode AC (a Mode A All Call, followed by a Mode C All Call)
  - The fixed doublet spacing for a Mode AC interrogation is the same as a Mode AA relative to each interrogating radio
- **W/S Sequence**
  - The MLAT Server commands only those W/S steps necessary to interrogate previously acquired targets



# ATCRBS Target Tracking

- **W/S Sequence (Targeted)**

- Series of range-limited interrogations with power levels specific to the targets in the region
- Analogous to selecting groups of darts on a dartboard by their ring number
- Sequence has higher repetition rate at the radio than the acquisition sequence



# Mode S

- **Mode S**

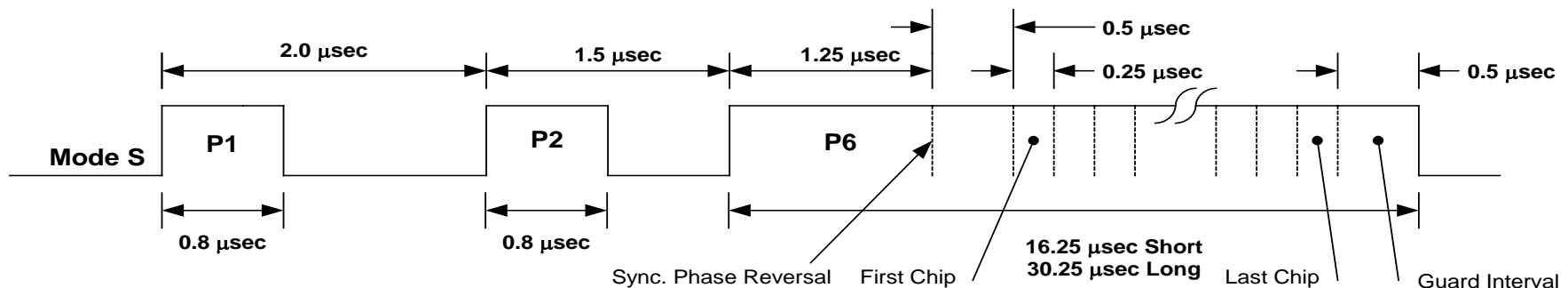
- Mode S transponder equipped aircraft assigned a unique ICAO 24-bit address code
- Mode S aircraft squitter their code once every second
- Using this unique code, interrogations can be directed to a specific aircraft and replies can be unambiguously identified
- Mode S uses a combination of “All-Call” and addressed interrogations.
  - WAM uses addressed Mode S interrogations
  - This permits replies from closely spaced aircraft to be received without mutual interference.
- Mode S signals can contain:
  - Mode S ID (always)
  - Aircraft identification information (Mode 3/A)
  - Altitude information (Mode C)
  - ADS-B



# Mode S

## • Mode S Interrogations

- Preamble consisting of P1 and P2 pulses (0.8  $\mu$ sec wide, 2  $\mu$ sec apart) followed by a 16.25 or 30.25  $\mu$ sec data block (1.5  $\mu$ sec after P2).
- P1 and P2 are used to suppress ATCRBS replies
- Data block is an RF pulse using phase reversals to represent binary data
  - Called Binary Differential Phase Shift Keying (DPSK) - potential phase reversals called “chips”
  - 180 degree phase reversal = binary 1
  - No phase reversal = binary 0
  - 56 or 112 data “chips”
  - Sync Phase (first phase reversal) 2.75  $\mu$ sec after P2
  - Data chips start 0.5  $\mu$ sec after sync phase and can occur every 0.25  $\mu$ sec
  - Guard Interval last 0.5  $\mu$ sec

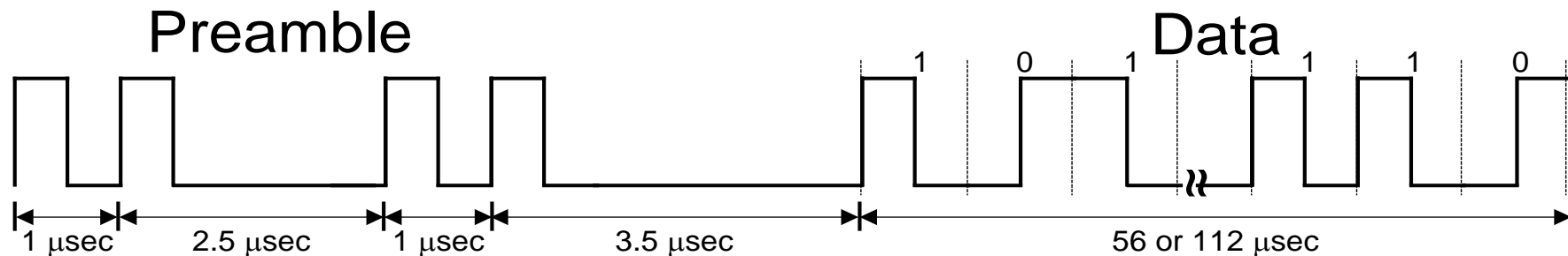




# Mode S

- **Mode S Replies**

- Mode S transponders reply with their address and any other information using a preamble and data block.
- Preamble consists of four 0.5  $\mu$ sec pulses separated by 1, 2.5, and 1  $\mu$ sec.
- 3.5  $\mu$ sec after the last pulse is a 56 or 112  $\mu$ sec data block.
- A data block is divided into 1  $\mu$ sec segments. Each segment is a bit.
- If the first half of the segment has a pulse, the bit is a 1. If the second half of the segment has a pulse, the bit is a 0.



# Mode S Uplink and Downlink Formats

<b>Uplink Format/Modes</b>	<b>Downlink Format/Modes</b>	<b>Content</b>
UF 0	DF 0	Short Air/Air Surveillance
UF 4	DF 4	Altitude Request/Altitude Reply
UF 4	DF 20	Altitude Request/Comm B Altitude Reply
UF 5	DF 5	ID Request/ID Reply
UF 5	DF 21	ID Request/Comm B ID Reply
UF 11	DF 11	Mode S Only All-Call/All-Call Reply
UF 16	DF 16	Long Air/Air Surveillance
UF 20	DF 4	Comm A Altitude Request/Altitude Reply
UF 20	DF 20	Comm A Altitude Request/Comm B Altitude Reply
UF 21	DF 5	Comm A ID Request/ID Reply
UF 21	DF 21	Comm A ID Request/Comm B ID Reply
UF 24	DF 24	Comm C/Comm D
	DF 17/18	ADS-B

Refer to RTCA DO-181E Section 2.2.14.1

Refer to ICAO Annex 10 Volume IV, July 2007, Sections 3.1.2.3.2.1.1 and 3.1.2.3.2.1.2



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# Mode-S Target Acquisition / Tracking

- **WAM utilizes spontaneous Mode S target squitters (DF11s) for target acquisition**
- **WAM performs TDOA calculations on multiple squitters to determine the horizontal position**
- **Flight ID and pressure altitude are determined by UF5/UF4 selective interrogations**
  - The position of the target is utilized to determine which radio is used for interrogations as well as the required power level
  - The rate at which interrogations are transmitted is determined by the update interval and data validity times required to support a given separation standard
- **Interrogations are not scheduled for 1090ES LV2 equipped aircraft.**



# ADS-B

- **ADS-B (Automatic Dependent Surveillance – Broadcast)**
  - Is currently transmitted and received using extended Mode S
  - Relies on satellite-based Global Positioning System (GPS)
  - Equipped aircraft broadcast their precise position along with other data, such as airspeed and altitude



# ADS-B Target Acquisition / Tracking

- **WAM utilizes spontaneous ADS-B target squitters (1090ES and UAT) for target acquisition**
- **WAM performs TDOA calculations on multiple squitters to determine the horizontal position**
- **Identification and pressure altitude is extracted from the received squitters**
  - No interrogations will be scheduled for these targets
- **Only ADS-B link Version 2 (260B/282B) targets are supported by this logic**
  - Link V2 is required for ADS-B Rule Compliance
  - Other link versions will require Mode S or ATCRBS interrogations to determine ID and pressure altitude



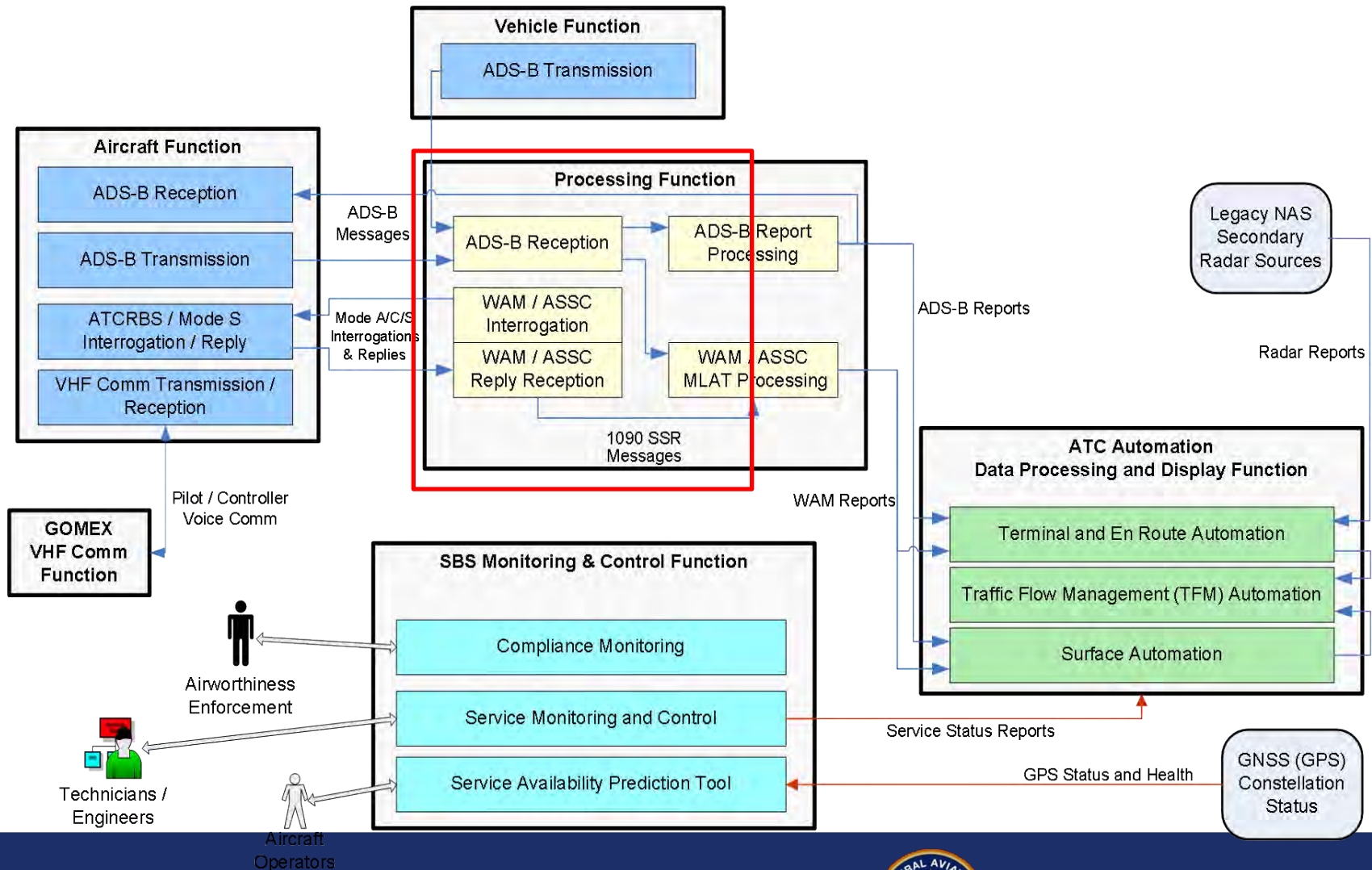


# Height Determination

- **MLAT normally uses 2-D Multilateration solution to make position estimates**
- **The altitude reported by the target is used as height in the MLAT solution**
  - 3-D Multilateration solution is used only when no Mode C is reported
- **Error between actual height and reported altitude can cause horizontal position bias**
  - Altitude correction is often used to minimize the errors
- **Uncorrected height is reported to automation systems**



# FAA Surveillance Functional Architecture



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# Conclusion

- **Multilateration Concept**
  - Based on Time Difference of Arrival (TDOA) Processing
  - Multiple RUs detect signal
  - WAM multilaterates on ATCRBS and Mode S
- **Time Difference of Arrival (TDOA)**
  - Time = Distance
  - RF energy travels 1 NM in 6.18  $\mu$ sec
  - Difference of signal arrival time at multiple RUs is used to generate solution arcs
  - Three RUs develop hyperbolic solution arcs
  - Four or more RUs develop hyperboloid solution arcs
- **ATCRBS and Mode S**
  - WAM interrogates ATCRBS using Whisper/Shout
  - WAM interrogates Mode S using addressed interrogations
- **Refer to Aeronautical Surveillance Manual DOC 9924**

